EAST SEARCH

9/24/2006

L#	Hits	Search String	Databases
S8	1	S7 and (throttle near2 setting)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S2	34	S1 and (turbocharger with (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S9	1	S7 and (throttle with setting)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S10	4	S7 and (model$3 with (turbocharger or (turbine near2 stage)))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S7	113	S2 or S3 or S6	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S3	49	S1 and (turbocharger same (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S6	113	S4 and S5	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S5	4854	S1 and (turbine near2 stage)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S4	1861	S1 and (turbocharger)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S1	102852	gas turbine or "jet engine" or (locomotive near2 "diesel engine")	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S11	118918	(gas near2 turbine) or (steam near2 turbine)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S12	2968	S11 and turbocharger	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S13	6536	S11 and (turbine near2 stage)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S14	148	S12 and S13	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S15	0	S14 and (throttle near2 setting)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S16	0	S14 and (throttle with setting)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S17	6	S14 and (throttle with position)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S18	102879	gas turbine or "jet engine" or (locomotive near2 "diesel engine")	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S19	34	S18 and (turbocharger with (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S20	49	S18 and (turbocharger same (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S27	150	S14 or S24	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S21	1861	S18 and (turbocharger)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S22	4854	S18 and (turbine near2 stage)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S23	113	S21 and S22	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S24	113	S19 or S20 or S23	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S25	1	S24 and (throttle with setting)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S26	6	S24 and (throttle with position)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S28	6	S27 and (model$3 with (turbocharger or (turbine near2 stage)))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S29	3	S27 and (model$3 with (blade or (nozzle near2 vane) or vane))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S30	13	S27 and (rotation near2 speed)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S31	29	S27 and (excitation or vibration)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S32	4	S27 and (natural near2 frequency)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S35	0	S27 and (vane near2 vibration)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S36	2	S27 and (fabricat$3 with (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S37	1	S27 and (harmonic with (excitation or vibration))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S33	2	S27 and (vane near2 excitation)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S34	3	S27 and (excitation near2 frequency)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S38	4	S27 and (fabricat$3 with turbocharger)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S39	13	S27 and (blade with (configuration or material or composition))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S40	14	S27 and (number with (vane or nozzle))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S41	2	S27 and (prime near2 number)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB

S42	86	S17 or S19 or S20 or S25 or S26 or S28 or S29 or S30 or S31 or S32 or S33 or S34 or S36 c	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S44	7	S42 and (S28 or S29)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S43	6	S42 and S26	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S45	13	S42 and S30	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S46	29	S42 and (S31 or S32 or S33 or S34 or S37)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S47	2	S42 and S41	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S48	4	S42 and S38	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S49	2	S42 and S36	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S50	26	S42 and (S39 or S40)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S51	96313	gas near2 turbine	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S52	4971	S51 and (turbine near2 stage)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S53	64	S52 and (throttle near2 (setting or position))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S54	20	S52 and (model$3 with (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S55	28	S52 and (model$3 with (blade or (nozzle near2 vane) or vane))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S56	238	S52 and (rotation near2 speed)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S57	474	S52 and (excitation or vibration)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S59	103	S52 and ((vane or blade) with (vibration or excitation))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S58	78	S52 and ((natural or resonan$2) near2 frequency)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S60	154	S52 and (fabricat$3 with ((turbine near2 stage) or turbine))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S61	6	S52 and (harmonic with (excitation or vibration))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S63	483	S52 and (blade with (configuration or material or composition))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S64	219	S52 and (blade with configuration)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S65	312	S52 and (blade with material)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S66	41	S52 and (blade with composition)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S67	341	S52 and (number with (vane or nozzle))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S62	25	S52 and (excitation with frequency)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S68	2	S52 and (prime near2 number)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S69	310	S53 or S54 or S55 or S58 or S59 or S61 or S62 or S66 or S68	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S70	1377	S56 or S57 or S60 or S64 or S65 or S67	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S71	210	S69 and S70	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S72	310	S69 or S71	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S73	96313	gas near2 turbine	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S74	4971	S73 and (turbine near2 stage)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S75	64	S74 and (throttle near2 (setting or position))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S76	20	S74 and (model$3 with (turbine near2 stage))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S77	28	S74 and (model$3 with (blade or (nozzle near2 vane) or vane))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S78	238	S74 and (rotation near2 speed)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S81	103	S74 and ((vane or blade) with (vibration or excitation))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S79	474	S74 and (excitation or vibration)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S80	78	S74 and ((natural or resonan$2) near2 frequency)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S82	154	S74 and (fabricat$3 with ((turbine near2 stage) or turbine))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S83	6	S74 and (harmonic with (excitation or vibration))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S84	25	S74 and (excitation with frequency)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S85	219	S74 and (blade with configuration)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S86	312	S74 and (blade with material)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S87	41	S74 and (blade with composition)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S88	341	S74 and (number with (vane or nozzle))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB

S89	2	S74 and (prime near2 number)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S91	1377	S78 or S79 or S82 or S85 or S86 or S88	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S92	210	S90 and S91	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S94	45	S93 and (S76 or S77)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S95	6	S93 and S83	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S96	2	S93 and S89	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S90	310	S75 or S76 or S77 or S80 or S81 or S83 or S84 or S87 or S89	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S93	310	S90 or S92	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S97	5	S74 and ((rotation near2 speed) with throttle)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S98	4	S93 and S97	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S99	79	S73 and ((rotation near2 speed) with throttle)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S101	3	S93 and ((diesel near2 engine) with turbocharger)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S102	10	S93 and ((combustion near2 engine) with turbocharger)	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S100	16	S73 and ((rotation near2 speed) with (throttle near2 (position or setting)))	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB
S103	12	S101 or S102	US-PGPUB; USPAT; USOCR; FPRS; EPO; JPO; DERWENT; IBM_TDB

10762396 **James Heilenbach et al.**

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9/24/2006

Results of search set S91:

Document Kind Codes	Title	Issue Date	Current OR	Abstract
US 20060184255 A1	Adaptive sensor model	20060817	700/44	
US 20060180420 A1	Vibration dampers	20060817	188/378	
US 20060177314 A1	Turbine rotor blade and turbine	20060810	416/219R	
US 20060127221 A1	Turbine moving blade	20060615	416/222	
US 20060126902 A1	Surface roughness measuring method and apparatus and turbine deterioration diagnostic me	20060615	382/108	
US 20060118215 A1	Precipitation hardened martensitic stainless steel, manufacturing method therefor, and turbin	20060608	148/607	
US 20060104818 A1	Blade	20060518	416/232	
US 20060096455 A1	APPARATUS AND PROCESS FOR POWER RECOVERY	20060511	95/269	
US 20060086090 A1	Vibration limiter for coaxial shafts and compound turbocharger using same	20060427	60/612	
US 20060086078 A1	Universal Carnot propulsion systems for turbo rocketry	20060427	60/226.1	
US 20060081701 A1	Method and apparatus for verifying connectivity of an instrumentation system	20060420	235/380	
US 20060078422 A1	Method for modifying gas turbine nozzle area	20060413	415/191	
US 20060067830 A1	Method to restore an airfoil leading edge	20060330	416/229R	
US 20060030450 A1	Hybrid vehicle formed by converting a conventional IC engine powered vehicle and method o	20060209	477/3	
US 20050274112 A1	Fatigue failure diagnostic method of turbocharger and fatigue failure diagnostic apparatus for	20051215	60/602	
US 20050254940 A1	Blade arrangement	20051117	415/170.1	
US 20050196278 A1	Turbine blade arrangement	20050908	416/97R	
US 20050194363 A1	MULTI-LASER BEAM WELDING HIGH STRENGTH SUPERALLOYS	20050908	219/121.64	
US 20050135932 A1	Turbine blade	20050623	416/97R	
US 20050126182 A1	Hybrid microturbine for generating electricity	20050616	60/791	
US 20050126171 A1	Uncoupled, thermal-compressor, gas-turbine engine	20050616	60/645	
US 20050111975 A1	Method for assembling gas turbine engine components	20050526	416/96R	
US 20050110991 A1	Methods and apparatus for evaluating rotary machinery	20050526	356/318	
US 20050103014 A1	Dual loop exhaust gas recirculation system for diesel engines and method of operation	20050519	60/605.2	

US 20050093214 A1	Spring mass damper system for turbine shrouds	20050505	267/136
US 20050084370 A1	Cooled turbine blade	20050421	416/95
US 20050074356 A1	Heat resisting steel, gas turbine using the steel, and components thereof	20050407	420/38
US 20050056313 A1	Method and apparatus for mixing fluids	20050317	137/3
US 20050042384 A1	Method of altering the frequency of blades for thermal fluid-flow machines	20050224	427/446
US 20050026095 A1	Multi-stage combustion using nitrogen-enriched air	20050203	431/2
US 20040225482 A1	Design and evaluation of actively cooled turbine components	20041111	703/2
US 20040219079 A1	Trifluid reactor	20041104	422/194
US 20040216458 A1	Electric motor assisted turbocharger	20041104	60/608
US 20040177618 A1	Methods for operating gas turbine engines	20040916	60/775
US 20040101402 A1	Turbine	20040527	415/160
US 20040093147 A1	Method and system for temperature estimation of gas turbine combustion cans	20040513	701/100
US 20040083731 A1	Uncoupled, thermal-compressor, gas-turbine engine	20040506	60/645
US 20040076540 A1	Welding material, gas turbine blade or nozzle and a method of repairing a gas turbine blade (	20040422	420/450
US 20040069069 A1	Probe for measuring parameters of a flowing fluid and/or multiphase mixture	20040415	73/736
US 20040060298 A1	Dynamically uncoupled can combustor	20040401	60/772
US 20040025491 A1	Gas turbine set	20040212	60/39.182
US 20040020206 A1	HEAT ENERGY UTILIZATION SYSTEM	20040205	60/670
US 20030228225 A1	Turbine bucket	20031211	416/235
US 20030215330 A1	Turbines and their components	20031120	415/191
US 20030205042 A1	OVERTHRUST PROTECTION SYSTEM AND METHOD	20031106	60/204
US 20030194320 A1	Method of fabricating a shape memory alloy damped structure	20031016	416/96A
US 20030193331 A1	Method for in-situ eddy current inspection of coated components in turbine engines	20031016	324/240
US 20030156942 A1	Blades having coolant channels lined with a shape memory alloy and an associated fabricatic	20030821	416/96R
US 20030152879 A1	Multi-stage combustion using nitrogen-enriched air	20030814	431/8
US 20030084656 A1	Gas turbine set	20030508	60/39.5
US 20030083827 A1	Methods and systems for performing integrated analyses, such as integrated analyses for gas	20030501	702/34
US 20030082053 A1	Repair of advanced gas turbine blades	20030501	416/224
US 20030065436 A1	Gas turbine and operation method of gas turbine combined electric generating plant, gas turb	20030403	701/100
US 20030039542 A1	Transition piece side sealing element and turbine assembly containing such seal	20030227	415/135
US 20030036865 A1	Methods and systems for managing resources, such as engineering test resources	20030220	702/81
US 20030033813 A1	Cycle gas turbine engine	20030220	60/774
US 20030007866 A1	Shroud integral type moving blade and split ring of gas turbine	20030109	415/182.1
US 20030002975 A1	COMBUSTOR HOT STREAK ALIGNMENT FOR GAS TURBINE ENGINE	20030102	415/1
US 20030000221 A1	High pressure gas cycle and power plant	20030102	60/776
US 20020189229 A1	Gas turbine for power generation and combined power generation system	20021219	60/39.182
US 20020136638 A1	PRE-SEGMENTED SQUEALER TIP FOR TURBINE BLADES	20020926	416/223A
US 20020121414 A1	Friction vibration damper	20020905	188/268
US 20020100281 A1	Damper arrangement for reducing combustion-chamber pulsations	20020801	60/725
US 20020047071 A1	Lifting platform with energy recovery	20020425	244/199.1
US 20020046560 A1	High pressure gas cycle and power plant	20020425	60/39.39
US 20010040062 A1	Lifting platform	20011115	180/117
US 7104757 B2	Cooled turbine blade	20060912	416/97R
US 7064825 B2	Methods and apparatus for evaluating rotary machinery	20060620	356/318
US 7048782 B1	Apparatus and process for power recovery	20060523	95/269
US 7021896 B2	Turbine blade	20060404	416/97R
US 7021892 B2	Method for assembling gas turbine engine components	20060404	415/115
US 7003940 B2	System for control and regulation of the flame temperature for single-shaft gas turbines	20060228	60/39.25

US 6988365 B2	Dual loop exhaust gas recirculation system for diesel engines and method of operation	20060124	60/605.2
US 6972390 B2	Multi-laser beam welding high strength superalloys	20051206	219/121.64
US 6957541 B2	Gas turbine and operation method of gas turbine combined electric generating plant, gas turb	20051025	60/782
US 6952639 B2	Method and system for temperature estimation of gas turbine combustion cans	20051004	701/100
US 6942203 B2	Spring mass damper system for turbine shrouds	20050913	267/160
US 6935119 B2	Methods for operating gas turbine engines	20050830	60/775
US 6932565 B2	Turbine	20050823	415/119
US 6908288 B2	Repair of advanced gas turbine blades	20050621	416/224
US 6886622 B2	Method of fabricating a shape memory alloy damped structure	20050503	164/98
US 6866092 B1	Two-phase heat-transfer systems	20050315	165/104.21
US 6846160 B2	Turbine bucket	20050125	416/190
US 6840048 B2	Dynamically uncoupled can combustor	20050111	60/772
US 6804612 B2	Methods and systems for performing integrated analyzes, such as integrated analyzes for gas	20041012	702/34
US 6802695 B2	Turbines and their components	20041012	416/223R
US 6802405 B2	Friction vibration damper	20041012	188/268
US 6796123 B2	Uncoupled, thermal-compressor, gas-turbine engine	20040928	60/520
US 6790030 B2	Multi-stage combustion using nitrogen-enriched air	20040914	431/8
US 6739839 B1	First-stage high pressure turbine bucket airfoil	20040525	416/223A
US 6736596 B2	Shroud integral type moving blade and split ring of gas turbine	20040518	415/173.1
US 6707297 B2	Method for in-situ eddy current inspection of coated components in turbine engines	20040316	324/240
US 6701717 B2	Cycle gas turbine engine	20040309	60/792
US 6699015 B2	Blades having coolant channels lined with a shape memory alloy and an associated fabricatic	20040302	416/96A
US 6655126 B2	Overthrust protection system	20031202	60/243
US 6644032 B1	Transition duct with enhanced profile optimization	20031111	60/752
US 6644012 B2	Gas turbine set	20031111	60/39.182
US 6632299 B1	Nickel-base superalloy for high temperature, high strain application	20031014	148/428
US 6632069 B1	Step of pressure of the steam and gas turbine with universal belt	20031014	415/173.5
US 6630113 B1	Methods and apparatus for treating waste	20031007	422/199
US 6616094 B2	Lifting platform	20030909	244/12.1
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US 6579066 B1	Turbine bucket	20030617	416/243
US 6574966 B2	Gas turbine for power generation	20030610	60/806
US 6565680 B1	Superalloy weld composition and repaired turbine engine component	20030520	148/428
US 6554562 B2	Combustor hot streak alignment for gas turbine engine	20030429	415/1
US 6553752 B2	High pressure gas cycle and power plant	20030429	60/39.38
US 6547049 B1	Particle vibration damper	20030415	188/379
US 6546729 B2	Damper arrangement for reducing combustion-chamber pulsations	20030415	60/725
US 6546713 B1	Gas turbine for power generation, and combined power generation system	20030415	60/39.182
US 6542859 B1	Method for designing a cyclic symmetric structure	20030401	703/7
US 6481197 B2	High pressure gas cycle and power plant	20021119	60/39.39
US 6478537 B2	Pre-segmented squealer tip for turbine blades	20021112	415/173.1
US 6468367 B1	Superalloy weld composition and repaired turbine engine component	20021022	148/428
US 6464459 B2	Lifting platform with energy recovery	20021015	415/208.2
US 6454156 B1	Method for closing core printout holes in superalloy gas turbine blades	20020924	228/165
US 6379110 B1	Passively driven acoustic jet controlling boundary layers	20020430	415/119
US 6358004 B1	Steam turbine power-generation plant and steam turbine	20020319	415/200
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US 6305078 B1	Method of making a turbine blade	20011023	29/889.7

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US 6301872 B1	High pressure gas cycle and power plant	20011016	60/772
US 6231307 B1	Impingement cooled airfoil tip	20010515	416/97R
US 6224334 B1	Steam turbine, rotor shaft thereof, and heat resisting steel	20010501	415/199.5
US 6220086 B1	Method for ascertaining surge pressure ratio in compressors for turbines	20010424	73/118.2
US 6215678 B1	Arc plasma-joule heated melter system for waste treatment and resource recovery	20010410	363/126
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US 6167693 B1	High pressure gas cycle and powder plant	20010102	60/39.38
US 6164055 A	Dynamically uncoupled low nox combustor with axial fuel staging in premixers	20001226	60/776
US 6162014 A	Turbine spline seal and turbine assembly containing such spline seal	20001219	415/170.1
US 6160238 A	Tunable molten oxide pool assisted plasma-melter vitrification systems	20001212	219/121.37
US 6146098 A	Tip shroud for cooled blade of gas turbine	20001114	416/97R
US 6129514 A	Steam turbine power-generation plant and steam turbine	20001010	415/200
US 6092989 A	Compressor for turbine and gas turbine	20000725	415/200
US 6074169 A	High and low pressure sides-integrating steam turbine, long blades thereof and combined cyc	20000613	416/241R
US 6066825 A	Methods and apparatus for low NO.sub.x emissions during the production of electricity from w	20000523	219/121.36
US 6062026 A	Turbocharging systems for internal combustion engines	20000516	60/605.2
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US 6018471 A	Methods and apparatus for treating waste	20000125	363/126
US 5983624 A	Power plant having a U-shaped combustion chamber with first and second reflecting surfaces	19991116	60/39.77
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US 5943866 A	Dynamically uncoupled low NOx combustor having multiple premixers with axial staging	19990831	60/737
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US 5511721 A	Braze blocking insert for liquid phase brazing operations	19960430	228/216
US 5507623 A	Alloy-coated gas turbine blade and manufacturing method thereof	19960416	416/241R
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US 5486091 A	Gas turbine airfoil clocking	19960123	415/194
US 5478207 A	Stable blade vibration damper for gas turbine engine	19951226	416/219R
US 5473882 A	Combustion apparatus for a gas turbine having separate combustion and vaporization zones	19951212	60/776
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US 5332358 A	Uncoupled seal support assembly	19940726	415/174.5
US 5321949 A	Staged fuel delivery system with secondary distribution valve	19940621	60/739
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US 5281097 A	Thermal control damper for turbine rotors	19940125	416/193A
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US 5222742 A	Seal arrangement	19930629	277/420
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US 4744726 A	Turboset with at least one low-pressure turbine stage having an outer housing and an inner h	19880517	415/199.4
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US 4215412 A	Real time performance monitoring of gas turbine engines	19800729	701/100
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US 4144768 A	Apparatus for analyzing complex acoustic fields within a duct	19790320	73/646
US 4132816 A	Gas phase deposition of aluminum using a complex aluminum halide of an alkali metal or an	19790102	427/237
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US 4059972 A	Turbine shaft balancing	19771129	464/23
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US 3973391 A	Control apparatus for modulating the inlet guide vanes of a gas turbine employed in a combin	19760810	60/794
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US 3846065 A	VAPOR GENERATORS WITH LOW POLLUTANT EMISSION	19741105	431/347
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US 3739580 A	PROPULSION SYSTEM CONTROL	19730619	60/204
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US 3300966 A	Control mechanism for adjustable gas turbine nozzle	19670131	60/791
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US 3169747 A	Rotary bladed power conversion machines	19650216	415/195
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US 3095030 A	Hydro-mechanical governor	19630625	60/243
US 3093968 A	Method and apparatus for augmenting the drive of a gas turbine	19630618	60/39.21
US 3056454 A	Fuel systems for propeller-driving gas turbine engines	19621002	416/29
US 3048014 A	Combustion chamber for jets and similar engines	19620807	60/39.23
US 3044262 A	Control mechanism for adjustable gas turbine nozzle	19620717	60/791
US 3018623 A	Explosion gas turbines	19620130	60/804
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US 2943839 A	Elastic fluid mechanism	19600705	415/192
US 2926494 A	Fuel control system	19600301	60/39.281
US 2857132 A	Turbine wheel	19581021	416/92
US 2743578 A	Turbojet engine control system	19560501	60/39.281
US 2709893 A	Gas turbine power plant with heat exchanger and cooling means	19550607	60/39.511
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US 2668006 A	Turbocharger	19540202	416/171
US 2618431 A	Control system for gas turbine air compressor plants	19521118	417/28
US 2617253 A	Safety control system for cooling a gas turbine power plant on shutdown	19521111	60/39.091
US 2575237 A	Multistage bladed rotor	19511113	416/201R
US 2565324 A	Gas turbine with throttling air turbine in compressor intake	19510821	60/795
US 2559623 A	Turbo-supercharger system having controlled inlet and exhaust waste gates	19510710	417/29
US 2540991 A	Gas reaction aircraft power plant	19510206	244/209
US 2500234 A	Compressor surge control for exhaust turbine driven superchargers	19500314	60/600
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